

Kinematics in rotating frames

24 Apr.

REVIEW: notation, fictitious forces
Foucault pendulum

BODY FRAME: UNPRIMED

SPACE FRAME: PRIMED

Vector: \vec{e} "abstract"

$$= v_i \hat{x}_i(t) \xrightarrow{\text{more gen.}} v_i(t) \hat{x}_i(t)$$

$$= \underbrace{v'_i(t)}_{\text{const vec in Body frame}} \hat{x}_i \xrightarrow{\text{GENERAL VECTOR}} v'_i(t) \hat{x}_i$$

still no t-dep

[apology for the different notations
btwn lecture & section]

ROTATION BETWEEN FRAMES: $e'_i = u_{ij} e_j$

TOTAL TIME RATE OF CHANGE:

$$\dot{e}'_i = \frac{d}{dt} (u_{ij} e_j) = \dot{u}_{ij} e_j + u_{ij} \dot{e}_j$$

$$= \underbrace{\dot{u}_{ij}}_{A'} \underbrace{e_j}_{e'} + \dot{e}'_i$$

$$\downarrow \qquad \qquad \downarrow \qquad \qquad \downarrow$$

$$e'_i \qquad \qquad = \vec{\omega} \times e'_i + \dot{e}'_i$$

↑
cf HW 8

OPEN DOT: the 'velocity' without the time evolution of the basis vectors

$$\dot{\mathbf{e}}_i = \dot{e}_x \hat{x} + \dot{e}_y \hat{y} + \dot{e}_z \hat{z}$$

↑
time deriv/velocity
that we obs as
people stuck on
earth's surface

⊗

Body frame expansion

↳ but BASIS VECTORS ALSO

TIME EVOLVE!

so $\dot{\mathbf{e}}_i$ is not the full time deriv.

cf.

$$\dot{\mathbf{e}}_i = \dot{e}'_x \hat{x}' + \dots \quad (\text{SPACE FRAME})$$

↑
these never time evolve
carries full time evolution

Q: CAN YOU WRITE $\dot{\mathbf{e}}_i$ IN SPACE FRAME?

in principle yes, but have to disentangle the basis evolution from body vector evolution. (why would this even be useful??)

Q: CAN YOU WRITE $\dot{\mathbf{e}}_i$ IN BODY FRAME?

$$\dot{\mathbf{e}}_i = \dot{e}_x \hat{x} + e_x \dot{\hat{x}} + \dots$$

So: $\dot{\vec{e}} = \vec{\omega} \times \vec{e} + \dot{\vec{e}}^0$ ← relates non-inertial frame physics to inertial frame laws

eg. $\vec{e} = \hat{x}$ (body frame)

↑
const. body frame components

$\Rightarrow \dot{\vec{e}}^0 = \dot{\hat{x}} = 0$

WHAT IS THE FULL TIME DERIVATIVE?

$\dot{\vec{e}} = \dot{\hat{x}} = \vec{\omega} \times \hat{x}$

↳ body axes precess about instantaneous angular velocity $\vec{\omega}$

eg. $\dot{\vec{\omega}} = \vec{\omega} \times \vec{\omega} + \dot{\vec{\omega}}^0 = \dot{\vec{\omega}}^0$

then: let $\vec{e} = \vec{v} = \vec{r} = \vec{\omega} \times \vec{r} + \dot{\vec{r}}^0$

$\dot{\vec{e}} = \dot{\vec{r}} = \vec{\omega} \times \vec{r} + \dot{\vec{r}}^0$
 $= \vec{\omega} \times \vec{r} + \dot{\vec{r}}^0$
 $= \vec{\omega} \times (\vec{\omega} \times \vec{r} + \dot{\vec{r}}^0) + (\vec{\omega} \times \vec{r} + \dot{\vec{r}}^0)^0$

↑
prop. ~~rule~~ RULE for 0?

cont'd

$$\begin{aligned}\dot{\vec{e}}_i &= \ddot{\vec{r}}_i = \vec{\omega} \times (\vec{\omega} \times \vec{r} + \dot{\vec{r}}) + \dot{\vec{\omega}} \times \vec{r} + \vec{\omega} \times \dot{\vec{r}} + \ddot{\vec{r}} \\ &= \ddot{\vec{r}} + \vec{\omega} \times (\vec{\omega} \times \vec{r}) + 2\vec{\omega} \times \dot{\vec{r}} + \dot{\vec{\omega}} \times \vec{r}\end{aligned}$$

multiply across by mass

[fun: bottle rocket on a rope - what if mass had time dependence?]

$$\vec{F} = \vec{F}_{\text{act}} + \underbrace{m\vec{\omega} \times (\vec{\omega} \times \vec{r}) + 2m\vec{\omega} \times \dot{\vec{r}} + m\dot{\vec{\omega}} \times \vec{r}}$$

↑
"FICTITIOUS FORCE"

BETTER NAME: NON INERTIAL FORCE

these guys are the
"fictitious" forces

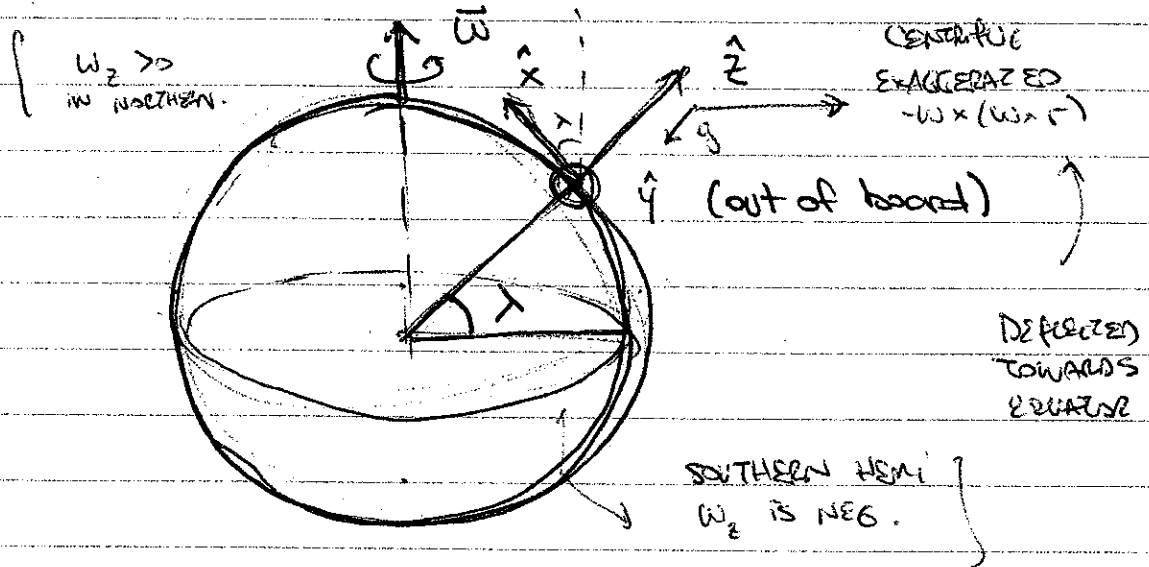
$$\vec{F}_{\text{act}} = \vec{F} - \underbrace{m\vec{\omega} \times (\vec{\omega} \times \vec{r})}_{\text{CENTRIFUGAL (}\propto \omega^2\text{)}} - \underbrace{2m\vec{\omega} \times \dot{\vec{r}}}_{\text{CORIOLIS}} - \underbrace{m\dot{\vec{\omega}} \times \vec{r}}_{\text{"EULER" (USUALLY IGNORED B/C } \dot{\vec{\omega}} = 0\text{)}}$$

↑
true force
(eg from V')

↑
CORIOLIS

↑
"EULER"
(USUALLY IGNORED
B/C $\dot{\vec{\omega}} = 0$)

Body axes: (latitude given by angle λ)



Q: This is the frame for the Northern Hemisphere ... how do we know that the dir of $\vec{\omega}$ is correct?

$$\vec{\omega} = \Omega (\cos \lambda \hat{x} + \sin \lambda \hat{z})$$

↑

$\frac{2\pi}{24 \text{ hrs}}$

↑

DIST. BETWEEN CIRCLES

↑

BODY COORDS

\hat{z} term is what gives \hat{x}, \hat{y} component of CORIOLIS FORCE $\sim \vec{\omega} \times \vec{v}$

↙ v in non int frame

Coriolis 'force': $-2m\vec{\omega} \times \vec{v}$

↑

SMALL ANGLE OSC. ASSUMP:

neglect v_z

$$= -2m\Omega [\cos \lambda v_y \hat{z} + \sin \lambda v_x \hat{y} - \sin \lambda v_y \hat{x}]$$

WE CARE ABOUT \hat{x} - \hat{y} PLANE MOTION

Gravitational force:

$$F_{x\text{grav}} = -m\omega_g^2 x$$

$$F_y = -m\omega_g^2 y$$

↑
 $\omega_g^2 = g/p$

Gravity + ^{Coriolis} ~~fictitious~~ 'force'

$$F_{ex} = -m\omega_g^2 x + 2m\Omega \sin \lambda \dot{y}$$

$$F_{ey} = -m\omega_g^2 y - 2m\Omega \sin \lambda \dot{x}$$

↑
 ω_p

then EOM:

$$\begin{cases} \ddot{x} = -\omega_g^2 x + 2\omega_p \dot{y} \\ \ddot{y} = -\omega_g^2 y - 2\omega_p \dot{x} \end{cases} (*)$$

↑
in BODY (EARTH) FRAME, THIS IS WHAT WE CARE ABOUT.

Usual 2ND ODE TRICK:

write $z = x + iy$, then $(*) \Rightarrow$

$$\ddot{z} = -\omega_g^2 z - i2\omega_p \dot{z}$$

Ansatz: $z = Ae^{i\alpha t}$

$$\Rightarrow -\alpha^2 = -\omega_g^2 + 2\omega_p \alpha$$

$$\uparrow \omega_g \gg \omega_p$$

$$\Rightarrow \alpha = \pm \omega_g + \mathcal{O}(\omega_p)$$

refine

$$-\alpha^2 = -\omega_g^2 \pm 2\omega_p \omega_g + \mathcal{O}(\omega_p^2)$$

$$\alpha^2 = \omega_g^2 (1 \mp 2\omega_p/\omega_g)$$

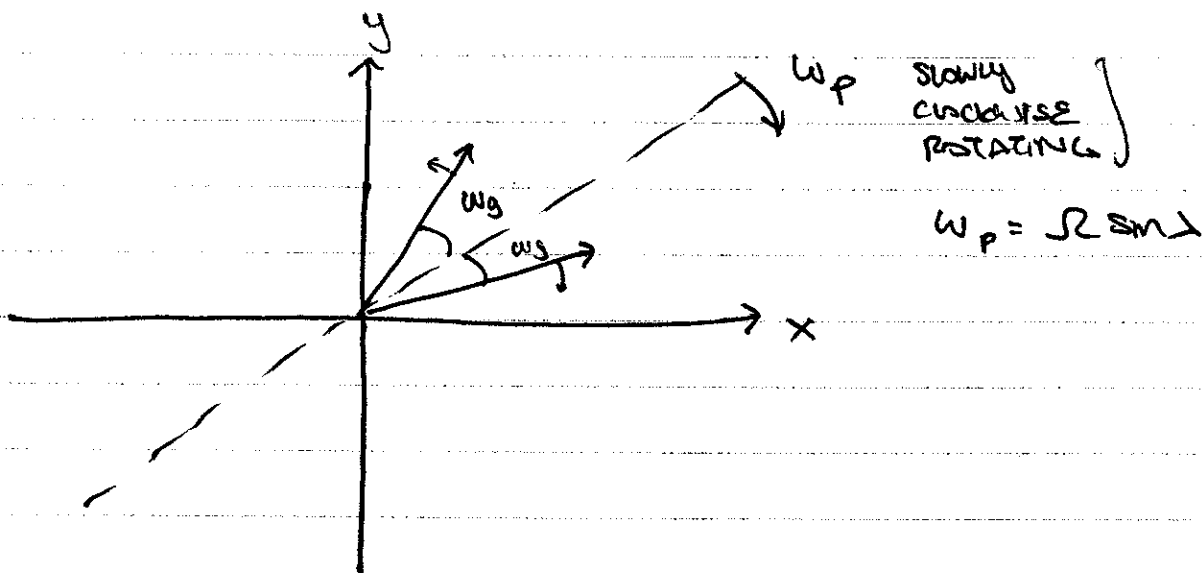
$$\alpha \approx \pm \omega_g (1 \mp \omega_p/\omega_g)$$

Approx sol's: $\alpha_{\pm} = \pm \omega_g - \omega_p$

$$z = A (e^{i(\omega_g - \omega_p)t} + e^{i(-\omega_g - \omega_p)t})$$

↑ + from init cond
eg. $(A, 0, 0)$ @ $t=0$

two counter rotating
phasors w/r to same
origin in phase



Another way to see (Marion & Thornton)

SUPPOSE $\omega_p = 0$: like a space frame
(no rotation of Earth)

$$\ddot{z}' = -\omega_g^2 z'$$

$$z = Ae^{i\omega t} + Be^{-i\omega t} \quad \leftarrow \text{USUAL MOTION}$$

COMPARE TO z ABOVE

$$z = e^{-i\omega_p t} z' \quad \leftarrow \text{USUAL MOTION}$$

$$x + iy = (\cos(\omega_p t) - i \sin(\omega_p t)) (x' + iy')$$

$$= (x' \cos \omega_p t + y' \sin \omega_p t) \leftarrow x$$
$$+ i(-x' \sin \omega_p t + y' \cos \omega_p t) \leftarrow y$$

$$\text{or: } \begin{pmatrix} x \\ y \end{pmatrix} = \begin{pmatrix} \cos \omega_p t & \sin \omega_p t \\ -\sin \omega_p t & \cos \omega_p t \end{pmatrix} \begin{pmatrix} x' \\ y' \end{pmatrix}$$

↑ ROTATION OF
PLANE OF PENDULUM